

# STiCM

## Select / Special Topics in Classical Mechanics

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STiCM Lecture 13: Unit 4

Symmetry and Conservation Laws  
Dynamical Symmetry in the Kepler Problem

Unit 1 : Equations of Motion (Newton / Lagrange / Hamilton )

Unit 2: Oscillators (Free/damped/driven), Resonances, Waves

Unit 3: Polar Coordinate Systems

Symmetry  $\longleftrightarrow$  Conservation laws (Noether)

Unit 4 : Dynamical Symmetry in the Kepler Problem

## Unit 4: Learning goals:

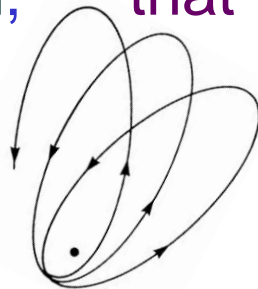
## Recapitulate:

**Conservation of energy** that is well-known in the Kepler-Bohr problem stems from the **symmetry with regard to temporal translations** (displacements on the time axis).

### GETTING CONSERVATION LAWS FROM THE EQUATION OF MOTION

Conservation of angular momentum, **likewise, stems from the central field symmetry in the Kepler problem.**

Neither of these accounts for the fact that the Kepler ellipse remains fixed; **that the ellipse does *not* undergo a 'rosette' motion.**





This unit will discover the ‘dynamical’ symmetry of the Kepler problem *and* its relation with the constancy (conservation) of the LRL vector, which keeps the orbit ‘fixed’.

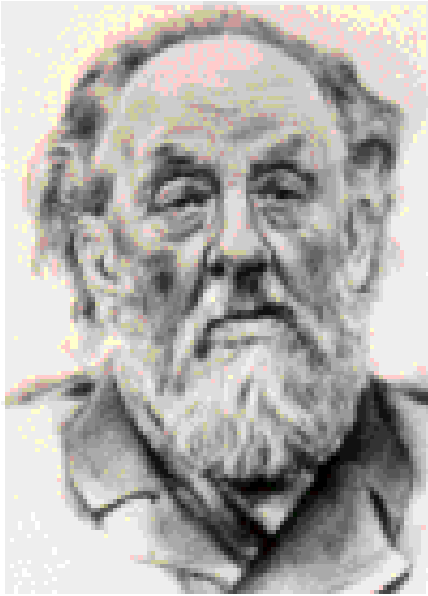
Motivation: Connection between symmetry & conservation laws has important consequences on issues at the very frontiers of physics and technology.

# Mechanics of Flights into Space

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# Mechanics of Flights into Space



Konstantin Tsiolkovsky  
(1857-1935)



Robert H. Goddard  
(1882-1945)



Hermann  
Oberth  
(1894-1989)

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Dr. Vikram  
Sarabhai

Father of India's  
Space Program

# Indian Space Research Organisation (ISRO)

## [1] Indian National Satellites (INSAT)

- for communication services

## [2] Indian Remote Sensing (IRS) Satellites

- for management of natural resources

## [3] Polar Satellite Launch Vehicle (PSLV)

- for launching IRS type of satellites

## [4] Geostationary Satellite Launch Vehicle (GSLV)

- for launching INSAT type of satellites.

Gravity plays the most important role in designing satellite trajectories,  
of course,  
and hence we study the Kepler TWO-BODY problem

We must then adapt the formalism to understand the models, methods and applications of satellite orbits, etc.

Other than 'energy' and 'angular momentum', what else is conserved, and what is the associated symmetry?

For given physical laws of nature, what quantities are conserved?

Rather, if you can observe what physical quantities are conserved, can you discover the physical laws of nature?



How did Kepler deduce that planetary orbits are ellipses around the sun ?



**Johannes Kepler**  
**1571- 1630**

*How would **you** solve this problem?*

Kepler had no knowledge of :  
(a) differential equations  
(b) inverse square force  
(gravity).

How did Kepler deduce that planetary orbits are ellipses around the sun ?

Kepler got his elliptic orbits not by solving differential equations for gravitational force, but by doing clever curve fitting of Tycho Brahe's experimental data.

## Tycho Brahe (1546-1601):

Danish astronomer appointed as the imperial astronomer under Rudolf II in Prague, which then came under the Roman empire.

Brahe had his own “Tychonian” model of planetary motion:

Brahe had planets revolve around the sun (like the Copernican system), and the sun and the moon going around the earth (like the system of Ptolemy).

What was Brahe’s **nose** was made of ?

## Tycho Brahe (1546-1601):

He discovered the supernova in 1572, in “Cassiopeia’.

Kepler and Brahe never could collaborate successfully.

They quarreled, and Tycho did not provide Kepler any access to the high precision observational data he (Brahe) had compiled.

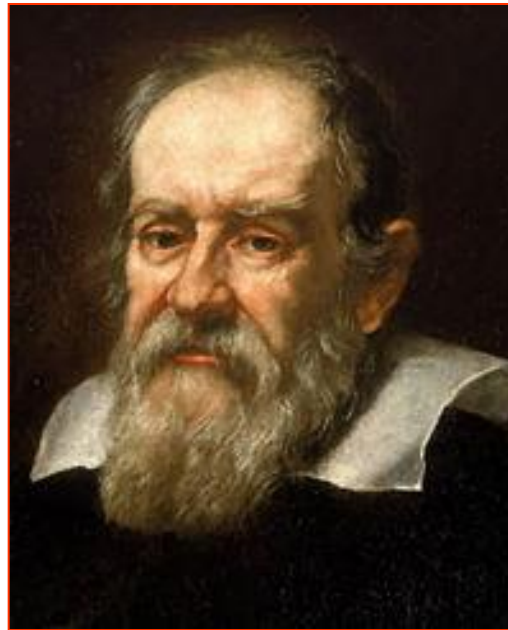
It was only on his deathbed, saying

*“.....let me not seem to have lived in vain.....”*  
that Brahe handed over his observational data to Kepler [Ref.:Sagan].

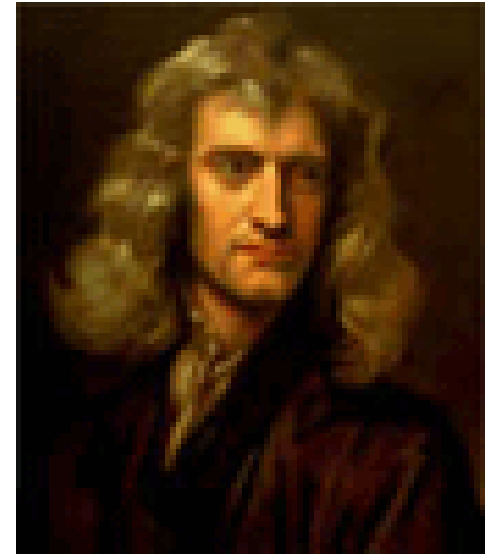
**Johannes Kepler**  
1571- 1630



**Galileo Galilei**  
1564 - 1642



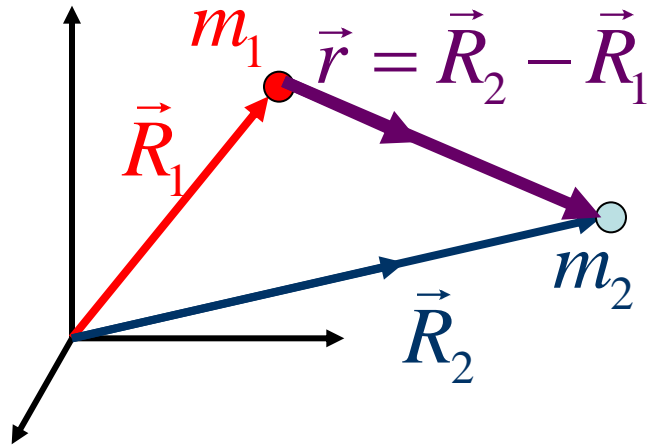
**Isaac Newton**  
(1642-1727)



Causality, Determinism, Equation of Motion

***'Dynamics' came well AFTER KEPLER!***

# Two-body problem



$$\hat{u} = \frac{\vec{r}}{|\vec{r}|}$$

$$\vec{F}_{\text{by\_1\_on\_2}}$$

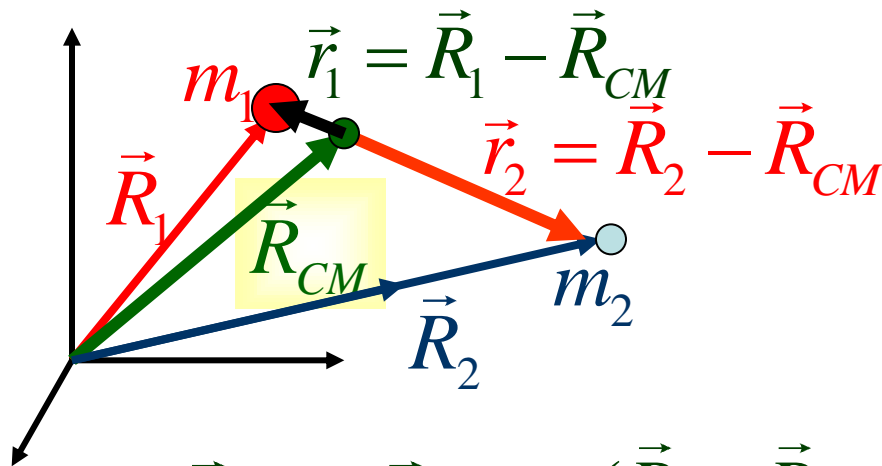
$$m_2 \ddot{\vec{R}}_2 = -G \frac{m_1 m_2}{|\vec{R}_2 - \vec{R}_1|^2} \hat{u}$$

$$\vec{F}_{\text{by\_2\_on\_1}}$$

$$m_1 \ddot{\vec{R}}_1 = G \frac{m_1 m_2}{|\vec{R}_2 - \vec{R}_1|^2} \hat{u}$$

# Two-body problem: Centre of Mass

$$\vec{R}_{CM} = \frac{m_1 \vec{R}_1 + m_2 \vec{R}_2}{m_1 + m_2}$$



$$m_1 \gg \gg m_2$$

$$\begin{aligned} m_1 \vec{r}_1 + m_2 \vec{r}_2 &= m_1 (\vec{R}_1 - \vec{R}_{CM}) + m_2 (\vec{R}_2 - \vec{R}_{CM}) \\ &= m_1 \vec{R}_1 + m_2 \vec{R}_2 - (m_1 + m_2) \vec{R}_{CM} \\ &= \vec{0}. \end{aligned}$$

$$\vec{r} = \vec{R}_2 - \vec{R}_1 = \vec{r}_2 - \vec{r}_1$$

$$\vec{F}_{\text{by}_1_{\text{on}}_2}$$

$$m_2 \ddot{\vec{R}}_2 = -G \frac{m_1 m_2}{|\vec{R}_2 - \vec{R}_1|^2} \hat{u}$$

$$\hat{u} = \frac{\vec{r}}{|\vec{r}|}$$

$$\vec{F}_{\text{by}_2_{\text{on}}_1}$$

$$m_1 \ddot{\vec{R}}_1 = G \frac{m_1 m_2}{|\vec{R}_2 - \vec{R}_1|^2} \hat{u}$$

$$\ddot{\vec{r}} = \ddot{\vec{R}}_2 - \ddot{\vec{R}}_1 = -G(m_1 + m_2) \frac{\vec{r}}{|\vec{r}|^3}$$

$$\vec{r} = \vec{R}_2 - \vec{R}_1 = \vec{r}_2 - \vec{r}_1$$

$$= -\kappa \frac{\vec{r}}{|\vec{r}|^3} \dots \text{where } \kappa = G(m_1 + m_2) \approx Gm_1 \dots \text{for } m_1 \gg m_2$$

$$\ddot{\vec{r}} + \kappa \frac{\vec{r}}{|\vec{r}|^3} = \vec{0}.$$

$$[\kappa] = L^3 T^{-2}$$

This equation of motion describes the 'relative motion' of the smaller mass relative to the larger mass, *assuming that the difference in the masses is huge.*



$$\ddot{\vec{r}} + \kappa \frac{\vec{r}}{|\vec{r}|^3} = \vec{0}.$$

We now take the dot product of the velocity with the 'Eq. of motion':

$$\dot{\vec{r}} \cdot \ddot{\vec{r}} + \frac{\kappa}{|\vec{r}|^3} \dot{\vec{r}} \cdot \vec{r} = 0$$

$$\dot{\vec{r}} = \frac{d}{dt} \vec{r} = \frac{d}{dt} (r \hat{u}) = \dot{r} \hat{u} + r \dot{\hat{u}}$$

$$\dot{\vec{r}} \cdot \ddot{\vec{r}} = \vec{v} \cdot \dot{\vec{v}} = v \dot{v}$$

$$\dot{\vec{r}} \cdot \vec{r} = (\dot{r} \hat{u} + r \dot{\hat{u}}) \cdot (r \hat{u}) = r \dot{r}$$

$$v \dot{v} + \frac{\kappa}{r^3} r \dot{r} = 0$$

$$v \lim_{\delta t \rightarrow 0} \frac{\delta v}{\delta t} = - \frac{\kappa \lim_{\delta t \rightarrow 0} \frac{\delta r}{\delta t}}{r^2}$$

$$\ddot{\vec{r}} + \kappa \frac{\vec{r}}{|\vec{r}|^3} = \vec{0}.$$

We took the dot product of the 'Eq. of motion' with velocity:

$$v \lim_{\delta t \rightarrow 0} \frac{\delta v}{\delta t} = - \frac{\kappa \lim_{\delta t \rightarrow 0} \frac{\delta r}{\delta t}}{r^2}$$

Integration  
w.r.t. time

Differentiation  
w.r.t. time

$$\frac{v^2}{2} = \frac{\kappa}{r} + E$$

$$\frac{v^2}{2} - \frac{\kappa}{r} = E$$

*DIMENSIONS*

$$[E] = L^2 T^{-2}$$

$$[\kappa] = L^3 T^{-2}$$

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Total 'SPECIFIC'  
(i.e. per unit mass)  
**MECHANICAL ENERGY:** Constant  
of integral /  
Constant of Motion.  
**INVARIANCE,  
SYMMETRY.**

$$\ddot{\vec{r}} + \kappa \frac{\vec{r}}{|\vec{r}|^3} = \vec{0}. \quad \text{Equation of Motion}$$

We took the dot product of the 'Eq. of motion' with velocity:

$$\dot{\vec{r}} \bullet \ddot{\vec{r}} + \frac{\kappa}{|\vec{r}|^3} \dot{\vec{r}} \bullet \vec{r} = 0$$

..... and discovered a CONSERVED QUANTITY!

$$\boxed{\frac{v^2}{2} - \frac{\kappa}{r} = E}$$

E: constant

symmetry with respect to translations in time.

(E,t): canonically conjugate pair of variables

We now take the cross product of the position vector with the 'Eq. of motion':

$$\vec{r} \times \ddot{\vec{r}} + \vec{r} \times \kappa \frac{\vec{r}}{|\vec{r}|^3} = \vec{0}.$$

*Therefore* :  $\vec{r} \times \ddot{\vec{r}} = \vec{0}$  Force: RADIAL  
central force symmetry

$$\text{Now, } \frac{d}{dt} (\vec{r} \times \dot{\vec{r}}) = \dot{\vec{r}} \times \dot{\vec{r}} + \vec{r} \times \ddot{\vec{r}} = \vec{r} \times \ddot{\vec{r}} = \vec{0}$$

*Therefore*  $\vec{H} = \vec{r} \times \dot{\vec{r}} = \vec{r} \times \vec{v}$  is also a constant of motion.

**SPECIFIC ANGULAR MOMENTUM**

**INVARIANCE, SYMMETRY**

**Homogeneity of time** ↔ **Conservation of energy**

**Homogeneity of space** ↔ **Conservation of linear momentum**  
**Translational Symmetry**

**Isotropy of space** ↔ **Conservation of angular momentum**  
**Rotational Symmetry**



**Emmy Noether**  
**1882 to 1935**

**SYMMETRY** ↔ **CONSERVATION LAWS**

Her entry to the Senate of the University of Gottingen, Germany, was resisted.

David Hilbert argued in favor of admitting her to the University Senate.

*Recapitulate...  
... From Unit 1*

$$\frac{\partial L}{\partial t} = 0$$

Time is homogeneous:  
Lagrangian of a closed system  
does not depend explicitly on time.

$$\left[ \dot{q} \frac{\partial L}{\partial \dot{q}} - L \right] \text{ is a CONSTANT.}$$

*Hamiltonian*  
**“ENERGY”**

**Conservation of Energy is  
thus connected with the  
symmetry principle  
regarding invariance with  
respect to temporal  
translations.**

## **Hamiltonian / Hamilton’s Principal Function**

## Space is homogenous and isotropic

the condition for homogeneity of space :  $\delta L(x, y, z) = 0$

*Recapitulate...  
... From Unit 1*

$$i.e., \delta L = \frac{\partial L}{\partial x} \delta x + \frac{\partial L}{\partial y} \delta y + \frac{\partial L}{\partial z} \delta z = 0$$

which implies  $\frac{\partial L}{\partial q} = 0$  where  $q = x, y, z$

since  $\frac{\partial L}{\partial q} - \frac{d}{dt} \left( \frac{\partial L}{\partial \dot{q}} \right) = 0$ , this means *i.e.*  $\frac{\partial L}{\partial \dot{q}} = p$  is conserved.

*i.e.*, is independent of time, is a constant of motion

**Law of conservation of momentum,  
arises from the homogeneity of space.**

**Symmetry**  $\longleftrightarrow$  **Conservation Laws**

*Recapitulate...  
... From Unit 1*

$$0 = \sum_{k=1}^N \vec{F}_k^{(i)} \cdot \overrightarrow{\delta S}$$

Displacement is arbitrary, hence,  $\sum_{k=1}^N \vec{F}_k^{(i)} = \vec{0}$ ;

$$i.e. \vec{0} = \frac{d}{dt} \sum_{k=1}^N \vec{p}_k^{(i)} = \frac{d}{dt} \vec{P}$$

Thus  $\vec{P}$  is conserved in the absence of external forces.

The conservation of momentum was secured in 'Unit 1' on the basis of 'translational invariance in homegenous space', and *not* on the basis of Newton's III law.



$$\frac{d\vec{P}}{dt} = \vec{0}$$
$$\frac{d\vec{p}_2}{dt} = -\frac{d\vec{p}_1}{dt}$$
$$\vec{F}_{12} = -\vec{F}_{21}$$

Given the symmetry related to translation in homogenous space, could you have discovered Newton's 3<sup>rd</sup> law?

Are the conservation principles consequences of the laws of nature? Or, are the laws of nature the consequences of the symmetry principles that govern them?

Until Einstein's special theory of relativity, it was believed that conservation principles are the result of the laws of nature.

Since Einstein's work, however, physicists began to analyze the conservation principles as consequences of certain underlying symmetry considerations, enabling the laws of nature to be revealed from this analysis.

Instead of introducing Newton's III law as a *fundamental principle*,  
*we deduced* it (in Unit 1) from symmetry / invariance.

This approach places SYMMETRY *ahead of* LAWS OF NATURE.

It is this approach that is of greatest value to contemporary physics. This approach has its origins in the works of Albert Einstein, Emmily Noether and Eugene Wigner.



(1879 – 1955)



(1882 – 1935)



(1902 – 1995)

$$\ddot{\vec{r}} + \kappa \frac{\vec{r}}{|\vec{r}|^3} = \vec{0}. \quad \text{Equation of Motion for Kepler's two-body problem}$$

$$\vec{H} = \vec{r} \times \dot{\vec{r}} = \vec{r} \times \vec{v},$$

the SPECIFIC ANGULAR MOMENTUM

We shall get yet another constant of motion, a conserved quantity, by taking the **cross product of the 'SPECIFIC ANGULAR MOMENTUM  $\vec{H}$ ' with the equation of motion:**

$$\vec{H} \times \ddot{\vec{r}} + \vec{H} \times \left( \kappa \frac{\vec{r}}{|\vec{r}|^3} \right) = \vec{0}.$$

$$\vec{H} \times \ddot{\vec{r}} + \vec{H} \times \left( \kappa \frac{\vec{r}}{|\vec{r}|^3} \right) = \vec{0}.$$

$$\vec{H} \times \ddot{\vec{r}} + \frac{\kappa}{|\vec{r}|^3} (\vec{r} \times \vec{v}) \times \vec{r} = \vec{0}$$

$$\vec{H} \times \ddot{\vec{r}} + \frac{\kappa}{|\vec{r}|^3} \{ (\vec{r} \cdot \vec{r}) \vec{v} - (\vec{r} \cdot \vec{v}) \vec{r} \} = \vec{0}$$

$$Use : \vec{r} \cdot \vec{v} = \vec{r} \cdot \frac{d}{dt} \{ r \hat{e}_\rho \} = \vec{r} \cdot \hat{e}_\rho \frac{dr}{dt} = r \dot{r}$$

$$\vec{H} \times \ddot{\vec{r}} + \frac{\kappa}{|\vec{r}|^3} (r^2 \vec{v} - r \dot{r} \vec{r}) = \vec{0}$$

$$\boxed{\vec{H} \times \ddot{\vec{r}}} + \frac{\kappa}{|\vec{r}|^3} (r^2 \vec{v} - r \dot{r} \vec{r}) = \vec{0}$$

Now,  $\frac{d}{dt}(\vec{H} \times \vec{v}) = \frac{d}{dt}(\vec{H} \times \dot{\vec{r}}) = \boxed{\vec{H} \times \ddot{\vec{r}}}$  .. since  $\dot{\vec{H}} = \vec{0}$

$$\frac{d}{dt}(\vec{H} \times \vec{v}) + \frac{\kappa}{|\vec{r}|^3} (r^2 \vec{v} - r^2 \dot{r} \hat{e}_\rho) = \vec{0}$$

$$\frac{d}{dt}(\vec{H} \times \vec{v}) + \kappa \frac{d}{dt} \left( \frac{\vec{r}}{r} \right) = \vec{0}$$

$$\frac{d}{dt}(\vec{H} \times \vec{v}) + \kappa \left( \frac{\vec{v}}{r} - \frac{\dot{r}}{r^2} \vec{r} \right) = \vec{0}$$

$$\frac{d}{dt} \left[ (\vec{H} \times \vec{v}) + \kappa \left( \frac{\vec{r}}{r} \right) \right] = \vec{0}$$

$$\frac{d}{dt} \left[ \left( \vec{H} \times \vec{v} \right) + \kappa \frac{\vec{r}}{r} \right] = \vec{0}$$

$$\left[ \left( \vec{H} \times \vec{v} \right) + \kappa \hat{e}_\rho \right] = -\vec{A},$$

constant

$$\vec{A} = \left( \vec{v} \times \vec{H} \right) - \kappa \hat{e}_\rho, \text{ constant}$$

LAPLACE – RUNGE – LENZ  
VECTOR

Physical  
Dimensions

$$[\kappa] = L^3 T^{-2}$$

$$\left[ \vec{v} \times \vec{H} \right]_{\text{PCD\_STICM}} = L T^{-1} \times L^2 T^{-1} = L^3 T^{-2}$$

Observe how a  
constant of  
motion has  
emerged –  
- yet again  
-- by playing with  
the equation of  
motion!

We will take a Break...

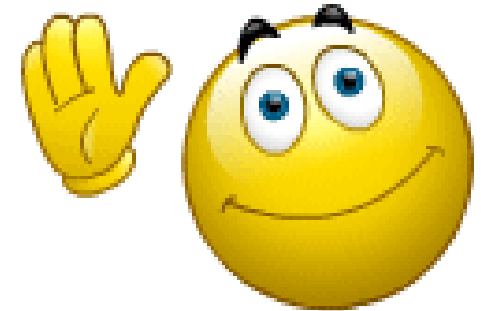
..... Any questions ?

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References:

[1] Oliver Montenbruck and Eberhard Gill  
'Satellite orbits – Models, Methods, Applications'.  
(Springer, Berlin, 2000)

[2] Francis J. Hale  
'Introduction to Space Flight'  
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**Next Lecture: Dynamical Symmetry  
of the Kepler Problem**



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## Select / Special Topics in Classical Mechanics

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**STiCM Lecture 14: Unit 4**

**Symmetry and Conservation Laws  
Dynamical Symmetry in the Kepler Problem**

$$\left( \vec{v} \times \vec{H} \right) - \kappa \hat{e}_\rho = \vec{A}, \text{ constant}$$

LAPLACE – RUNGE – LENZ VECTOR

Take dot product with  $\vec{r}$  :

→ Equation to the orbit / trajectory

→ Without solving the differential equation of motion!

$$\left( \vec{v} \times \vec{H} \right) - \kappa \hat{e}_\rho = \vec{A} \quad \text{where } \vec{H} = \vec{r} \times \vec{v}$$

Take dot product with  $\vec{r}$ :

$$\left( \vec{v} \times \vec{H} \right) \cdot \vec{r} - \kappa r = \vec{A} \cdot \vec{r}$$

*sign reversal*:  $\left( \vec{H} \times \vec{v} \right) \cdot \vec{r} + \kappa r = -\vec{A} \cdot \vec{r}$

Interchange 'dot' and 'cross':  $\vec{H} \cdot \left( \vec{v} \times \vec{r} \right) + \kappa r = -\vec{A} \cdot \vec{r}$

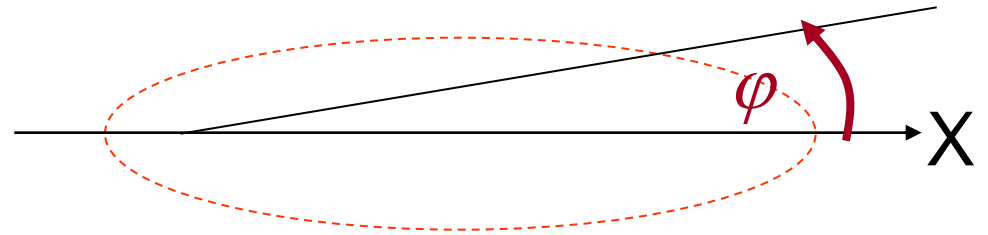
$$-H^2 + \kappa r = -\vec{A} \cdot \vec{r} = -Ar \cos \varphi$$

$$\varphi = \angle \left( \vec{A}, \vec{r}_{\text{CD\_STICM}} \right)$$

$$-H^2 + \kappa r = -Ar \cos \varphi$$

$$\varphi = \angle(\vec{A}, \vec{r})$$

$$\varphi = \angle(\vec{A}, \vec{r})$$

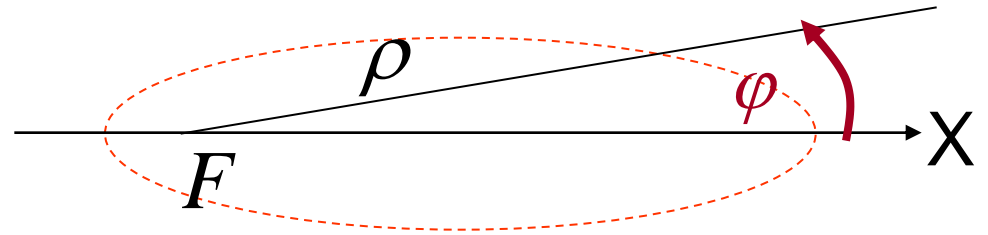


$$r(\kappa + A \cos \varphi) = H^2$$

$$r = \frac{H^2}{\kappa + A \cos \varphi} = \frac{\left(\frac{H^2}{\kappa}\right)}{1 + \left(\frac{A}{\kappa}\right) \cos \varphi} = \frac{p}{1 + \varepsilon \cos \varphi}$$

PCD\_STICM What is  $\varepsilon = \left(\frac{A}{\kappa}\right)$ ?

$$\varphi = \angle(\vec{A}, \vec{r})$$



$$\left[ (\vec{v} \times \vec{H}) - \kappa \left( \frac{\vec{r}}{r} \right) \right] = \vec{A}$$

$$\vec{H} = \vec{r} \times \vec{v} \Rightarrow \angle(\vec{H}, \vec{r}) = 90^\circ \text{ \& } \angle(\vec{H}, \vec{v}) = 90^\circ$$

$$\vec{H} \times \vec{v} = H v \hat{u}$$

$$(Hv)^2 + \frac{2\kappa}{r} (\vec{H} \times \vec{v}) \cdot \vec{r} + \kappa^2 = A^2$$

$$(\vec{H} \times \vec{v}) = (\vec{r} \times \vec{v}) \times \vec{v} = (\vec{v} \cdot \vec{r}) \vec{v} - v^2 \vec{r}$$

$$(\vec{H} \times \vec{v}) \cdot \vec{r} = (\vec{v} \cdot \vec{r}) \vec{v} \cdot \vec{r} - v^2 \vec{r} \cdot \vec{r} = (vr \cos \xi)^2 - v^2 r^2$$

$$= -v^2 r^2 \sin^2 \xi$$

$$\xi = \angle(\vec{r}, \vec{v})$$

$$H^2 v^2 - \frac{2\kappa}{r} v^2 r^2 \sin^2 \xi + \kappa^2 = A^2$$

PCD\_STiCM

$$H^2 v^2 - \frac{2\kappa}{r} v^2 r^2 \sin^2 \xi + \kappa^2 = A^2$$

$$H^2 \left[ v^2 - \frac{2\kappa}{rH^2} v^2 r^2 \sin^2 \xi \right] + \kappa^2 = A^2$$

$$\vec{H} = \vec{r} \times \vec{v}$$

$$H^2 \left[ v^2 - \frac{2\kappa}{r} \right] + \kappa^2 = A^2$$

$$H^2 [2E] + \kappa^2 = A^2$$

$$\frac{H^2 [2E]}{\kappa^2} + 1 = \frac{A^2}{\kappa^2}$$

$$\frac{A}{\kappa} = \sqrt{\frac{2EH^2}{\kappa^2} + 1}$$

$$r = \frac{p}{1 + \varepsilon \cos \varphi}$$

*with*

$$p = \frac{H^2}{\kappa} \quad \& \quad \varepsilon = \frac{A}{\kappa} = \sqrt{1 + \frac{2EH^2}{\kappa^2}}$$



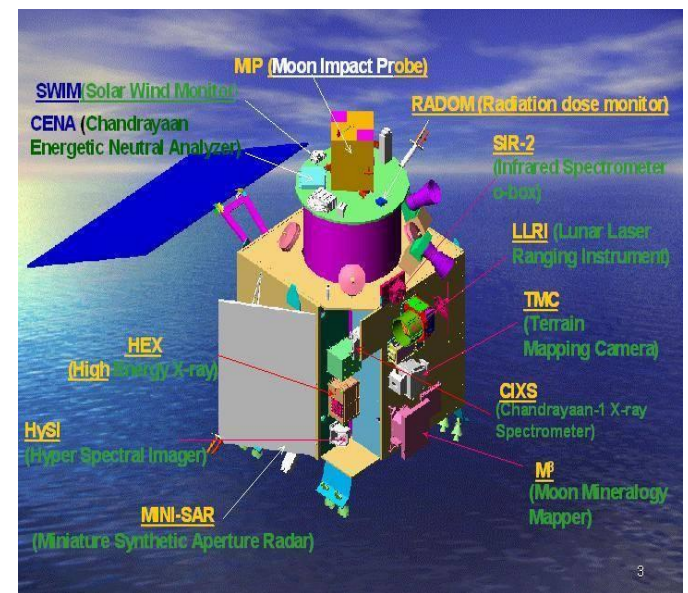
$$r = \frac{p}{1 + \varepsilon \cos \varphi}$$

$\varepsilon > 1$ : Hyperbola (open trajectory)

$\varepsilon = 1$ : Parabola (open trajectory)

$0 < \varepsilon < 1$ : Ellipse (closed trajectory)

$\varepsilon = 0$ : Circle  $\rightarrow$  degenerate ellipse



PCD\_STICM

For satellite and ballistic missile trajectories, ellipses (inclusive of the circle) are of primary interest.

$$r = \frac{p}{1 + \varepsilon \cos \varphi}$$

$\varepsilon > 1$ : Hyperbola (open trajectory)

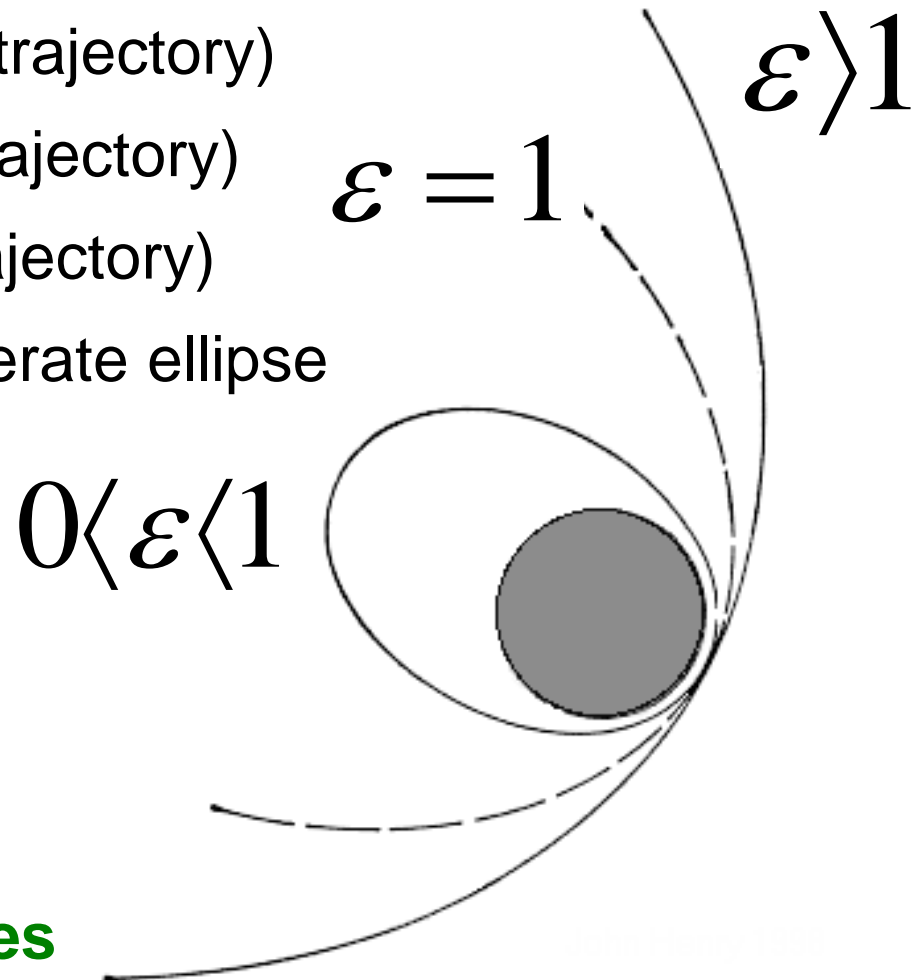
$\varepsilon = 1$ : Parabola (open trajectory)

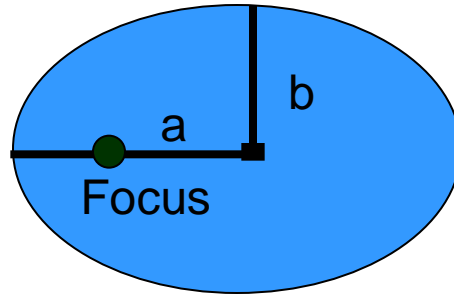
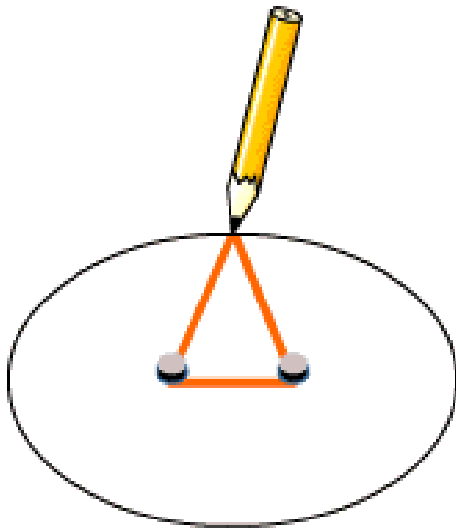
$0 < \varepsilon < 1$ : Ellipse (closed trajectory)

$\varepsilon = 0$ : Circle  $\rightarrow$  degenerate ellipse

**Earth-orbiting  
satellites are in  
elliptic motion.**

**Deep space probes  
leave earth's gravity  
on hyperbolic orbits**

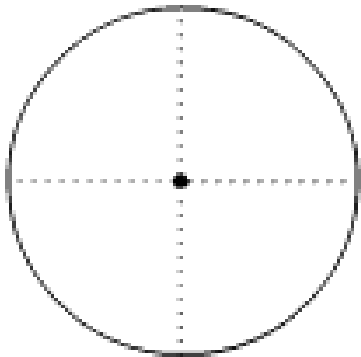
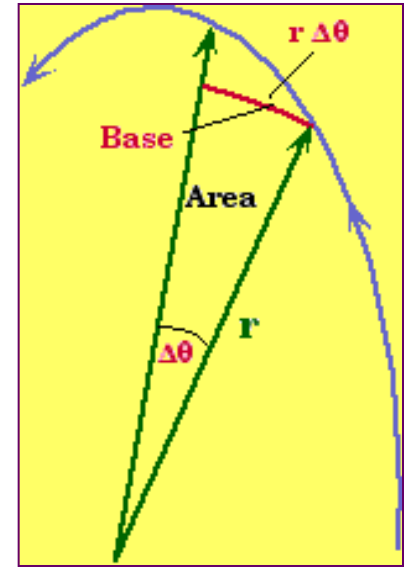




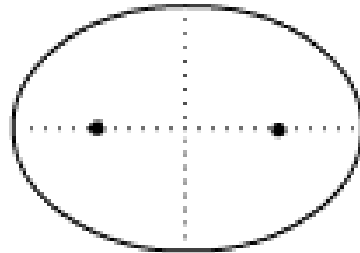
$a =$  semi-major axis

$b =$  semi-minor axis

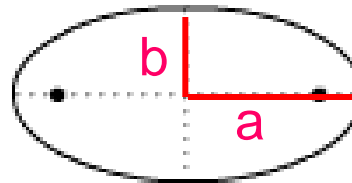
$e : \text{eccentricity}$   $\frac{b^2}{a^2} = 1 - e$



$e=0$



$e=0.5$



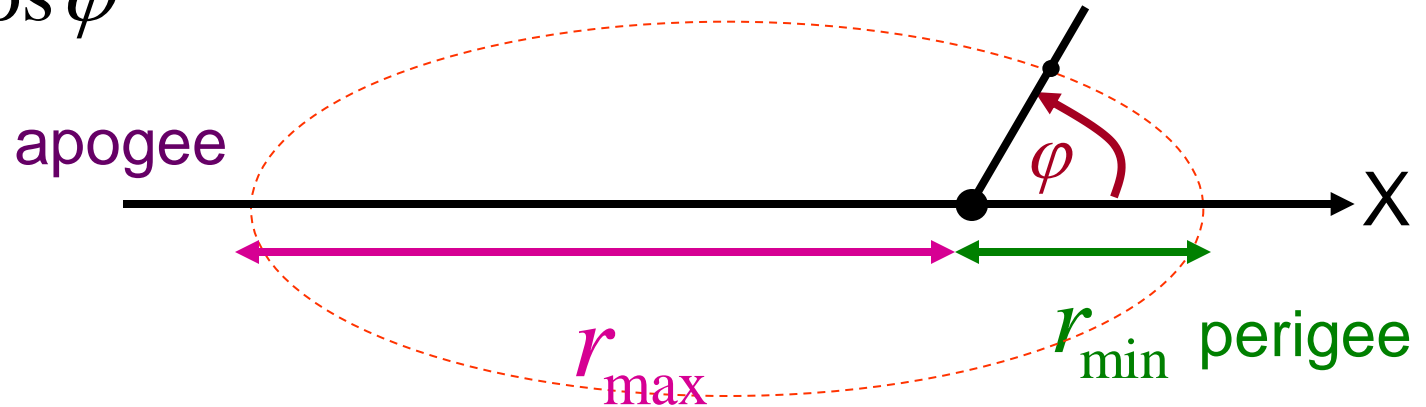
$e=0.75$



$e=0.95$

$$r = \frac{p}{1 + \varepsilon \cos \varphi};$$

$$0 \leq \varepsilon < 1$$

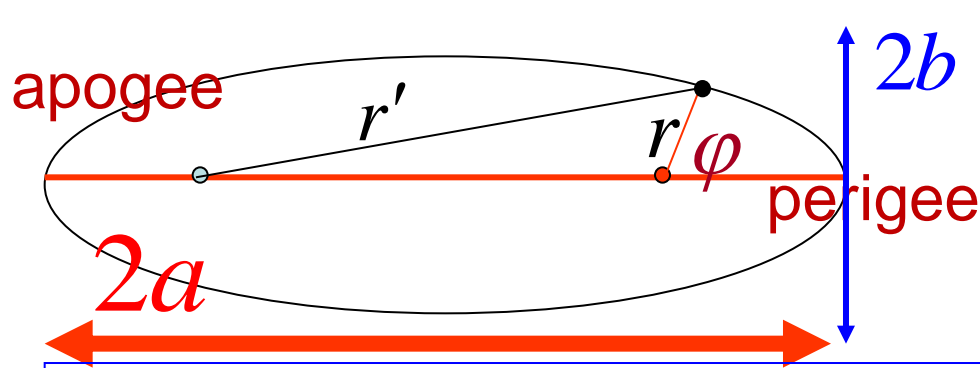


$$p = \frac{H^2}{\kappa} \quad \& \quad \varepsilon = \frac{A}{\kappa} = \sqrt{1 + \frac{2EH^2}{\kappa^2}}$$

perigee  $r = r_{\min}$  when  $\varphi = 0$ :  $r_{\min} = \frac{p}{1 + \varepsilon}$

apogee  $r = r_{\max}$  when  $\varphi = \pi$ :  $r_{\max} = \frac{p}{1 - \varepsilon}$

$$r = \frac{p}{1 + \varepsilon \cos \varphi}; \quad p = \frac{H^2}{\kappa}; \quad \varepsilon = \sqrt{1 + \frac{2EH^2}{\kappa^2}}$$



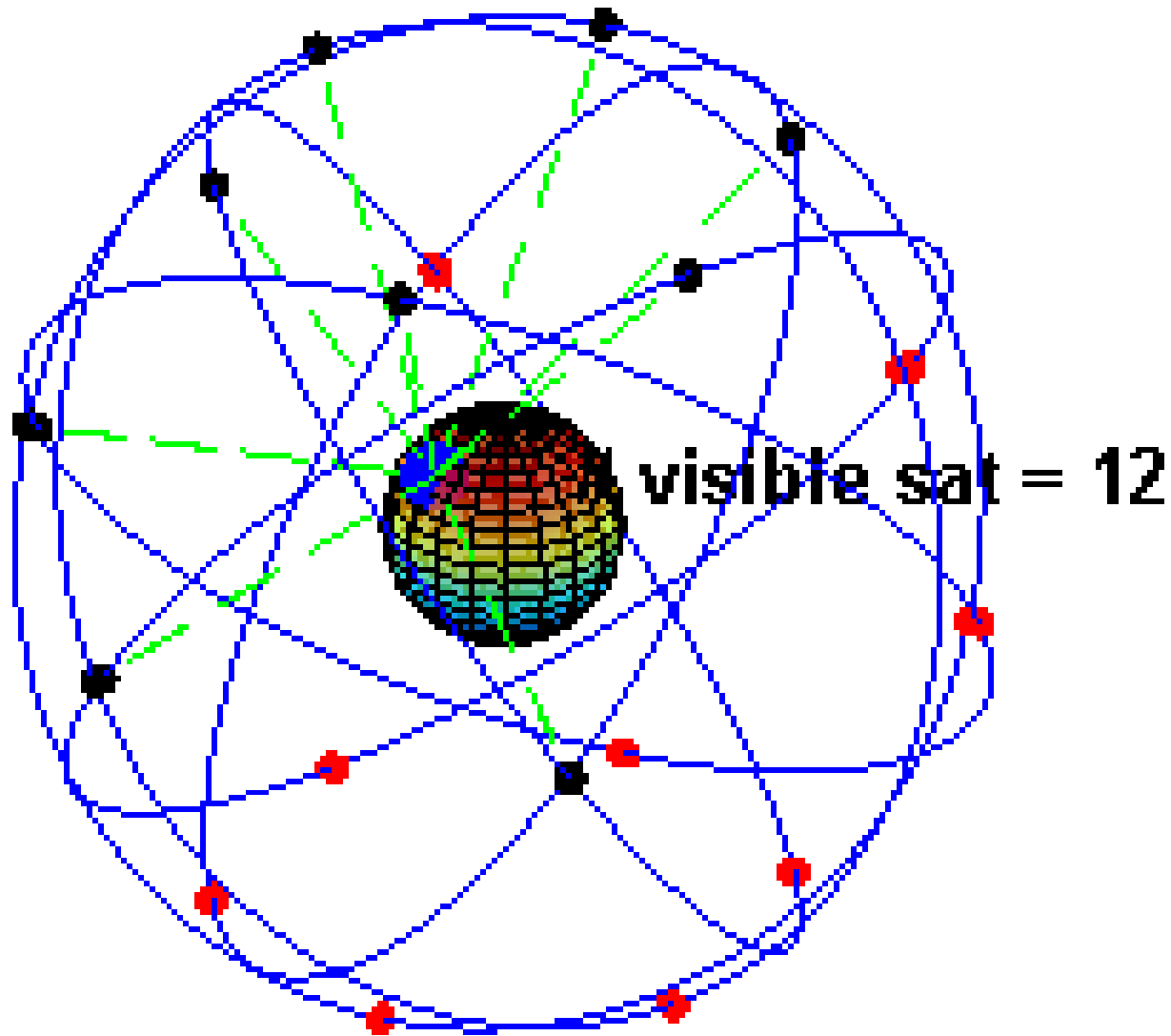
At apogee/perigee:  
 $r + r' = 2a$ , constant

$$2a = r_{\text{perigee}} + r_{\text{apogee}} = \frac{p}{1 - \varepsilon} + \frac{p}{1 + \varepsilon} = \frac{2p}{1 - \varepsilon^2}$$

$$p = a(1 - \varepsilon^2)$$

$$r = \frac{a(1 - \varepsilon^2)}{1 + \varepsilon \cos \varphi}$$

$$E = \frac{\kappa^2 (\varepsilon^2 - 1)}{2H^2} = \frac{\kappa (\varepsilon^2 - 1)}{2p} \xrightarrow{\text{for circular orbit}} \frac{-\kappa}{2a}$$



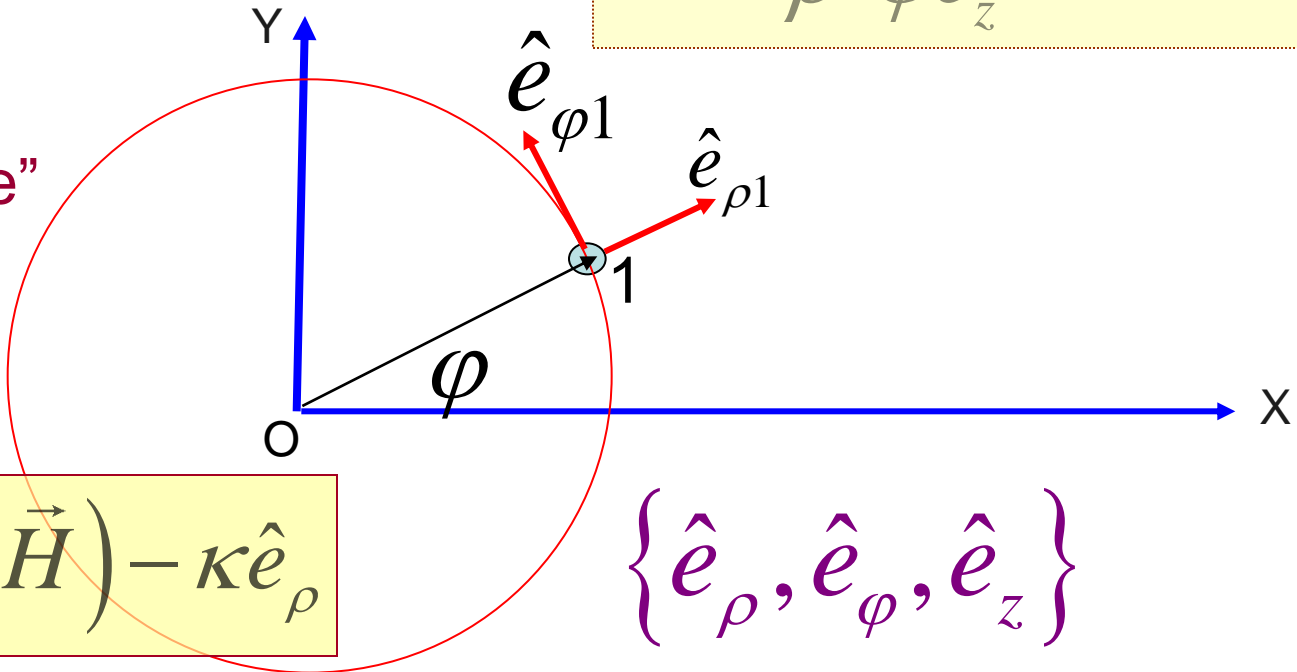
24 GPS satellites ---- Wikimedia Commons  
<http://en.wikipedia.org/wiki/File:ConstellationGPS.gif>

$$\vec{\rho} = \rho \hat{e}_\rho$$

$$\vec{v} = \dot{\rho} \hat{e}_\rho + \rho \dot{\varphi} \hat{e}_\varphi$$

$$\begin{aligned} \vec{H} &= \vec{r} \times \vec{v} \\ &= \rho \hat{e}_\rho \times (\dot{\rho} \hat{e}_\rho + \rho \dot{\varphi} \hat{e}_\varphi) \\ &= \rho^2 \dot{\varphi} \hat{e}_z \end{aligned}$$

“Unit Circle”



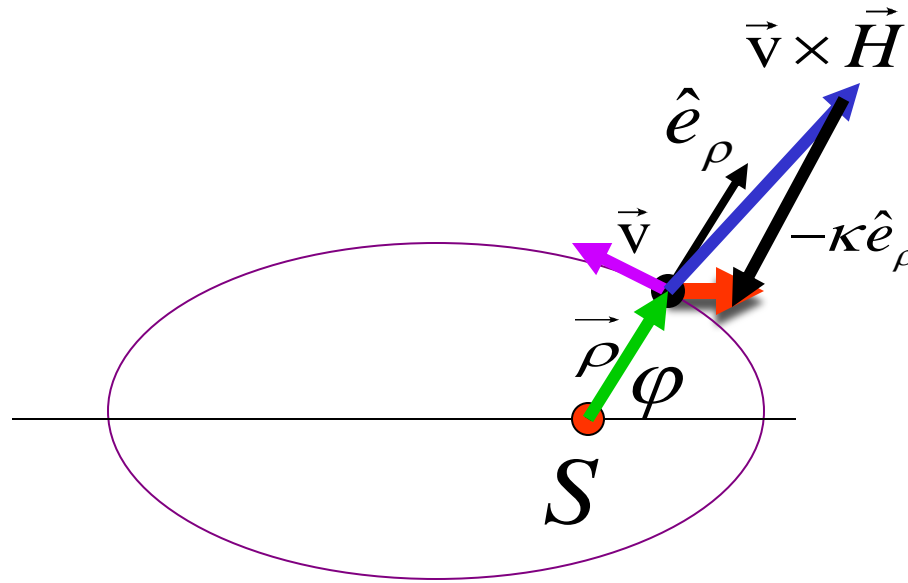
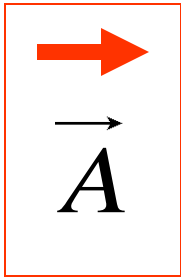
$$\vec{A} = (\vec{v} \times \vec{H}) - \kappa \hat{e}_\rho$$

$$\{\hat{e}_\rho, \hat{e}_\varphi, \hat{e}_z\}$$

## Plane/Cylindrical Polar Coordinate System

# Laplace Runge Lenz Vector

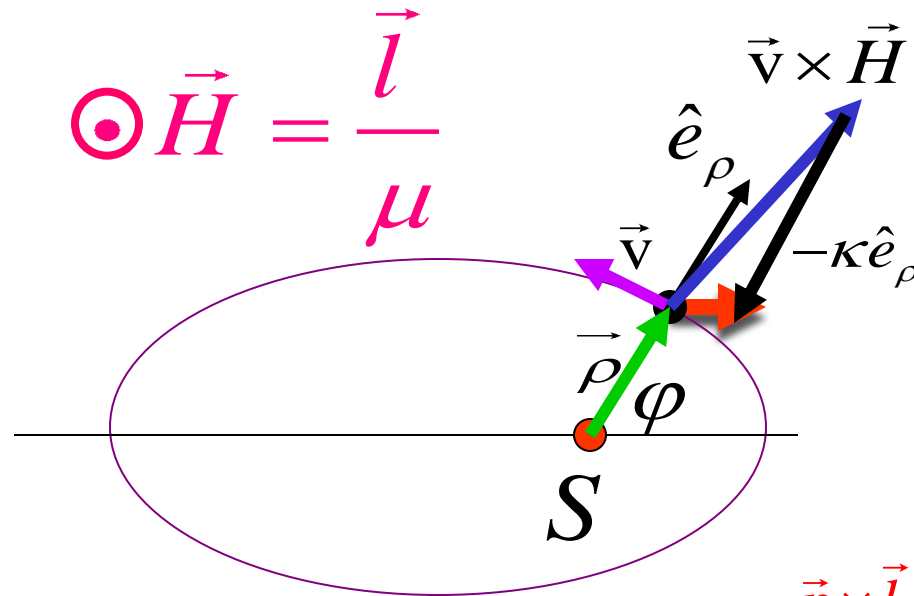
is constant for a strict  $\frac{1}{r}$  potential.



$$\vec{A} = \left( \vec{v} \times \vec{H} \right) - \kappa \hat{e}_\rho$$

The  $\odot \vec{H}$   
(specific)  
angular  
momentum  
vector is  
out of the  
plane of  
this figure,  
toward us.





The Laplace - Runge - Lenz vector :  $\vec{A} = \frac{\vec{p} \times \vec{l}}{\mu} - \mu \kappa \hat{e}_\rho$ ,

or alternatively defined in terms

of the specific angular momentum  $\vec{H}$  as

$$\vec{A} = \vec{v} \times \vec{H} - \kappa \hat{e}_\rho$$

$$\vec{A} = (\vec{v} \times \vec{H}) - \kappa \hat{e}_\rho$$

$$\frac{d\vec{A}}{dt} = \left( \frac{d\vec{v}}{dt} \times \vec{H} + \vec{v} \times \frac{d\vec{H}}{dt} \right) - \kappa \frac{d\hat{e}_\rho}{dt}$$

Central Field Symmetry  
Angular Momentum  
is Conserved

$$\frac{d\vec{A}}{dt} = \left( \frac{d\vec{v}}{dt} \times \vec{H} \right) - \kappa \frac{d\hat{e}_\rho}{dt}$$

But  $\frac{d\hat{e}_\rho}{dt} = \frac{\partial \hat{e}_\rho}{\partial \varphi} \dot{\varphi} = \hat{e}_\varphi \dot{\varphi}$

$$\frac{d\vec{A}}{dt} = \left( \frac{d\vec{v}}{dt} \times \vec{H} \right) - \kappa \dot{\varphi} \hat{e}_\varphi$$

$$\vec{A} = (\vec{v} \times \vec{H}) - \kappa \hat{e}_\rho$$

$$\frac{d\vec{A}}{dt} = \left( \frac{d\vec{v}}{dt} \times \vec{H} \right) - \kappa \dot{\varphi} \hat{e}_\varphi$$

What is the form of the force per unit mass:  $\frac{d\vec{v}}{dt}$  ?

We must know the form of the interaction!

$$\vec{F} = m\vec{a} = m \frac{d\vec{v}}{dt}$$

$$\frac{v^2}{2} - \frac{\kappa}{r} = E$$

The force per unit mass:  $\frac{d\vec{v}}{dt} = -\frac{\kappa}{\rho^2} \hat{e}_\rho$

$$\frac{d\vec{A}}{dt} = \left( \frac{d\vec{v}}{dt} \times \vec{H} \right) - \kappa \dot{\varphi} \hat{e}_\varphi$$

$$\frac{d\vec{A}}{dt} = \left( -\frac{\kappa}{\rho^2} \hat{e}_\rho \times (\rho^2 \dot{\varphi} \hat{e}_z) \right) - \kappa \dot{\varphi} \hat{e}_\varphi$$

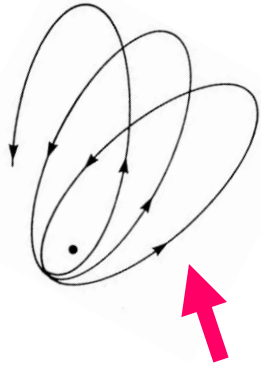
$\{\hat{e}_\rho, \hat{e}_\varphi, \hat{e}_z\}$ : right handed basis set

$$-\hat{e}_\rho \times \hat{e}_z = \hat{e}_\varphi$$

$$\frac{d\vec{A}}{dt} = \vec{0}$$



Rosette motion



Angular momentum is conserved,  
but major-axis not fixed.



Two-body central field Kepler-Bohr problem,

Attractive force:  
inverse-square-law.



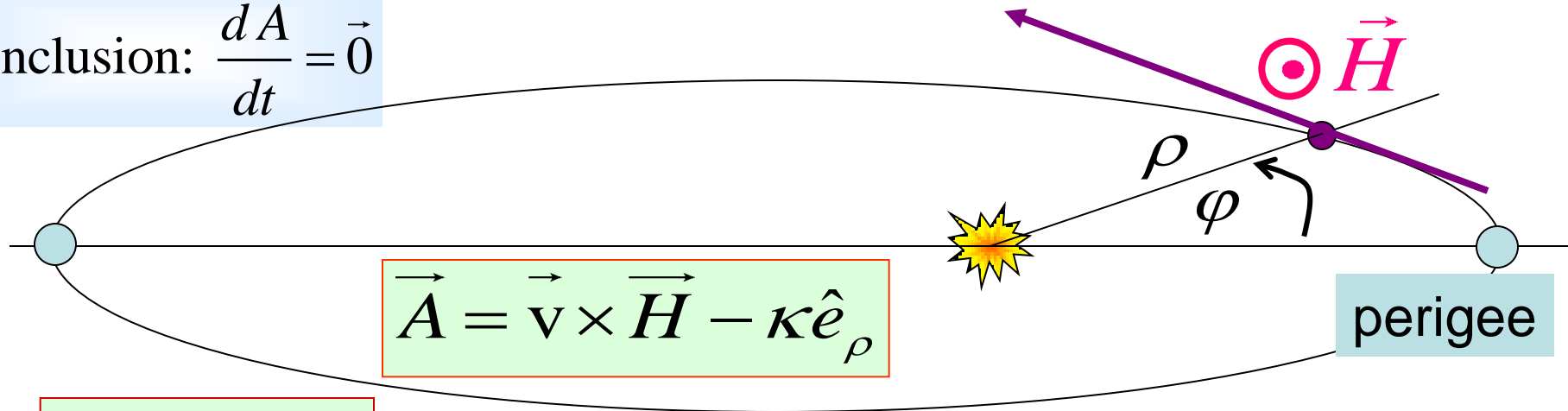
For this potential and for the associated field:

no precession of orbit.

The constancy of the orbit suggests a conserved quantity and one must look for an

associated symmetry.

Conclusion:  $\frac{d\vec{A}}{dt} = \vec{0}$

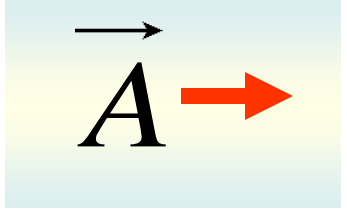


$$\vec{A} = \vec{v} \times \vec{H} - \kappa \hat{e}_\rho$$

$$\vec{A} \cdot \vec{H} = 0$$

LRL vector is in the plane of the orbit

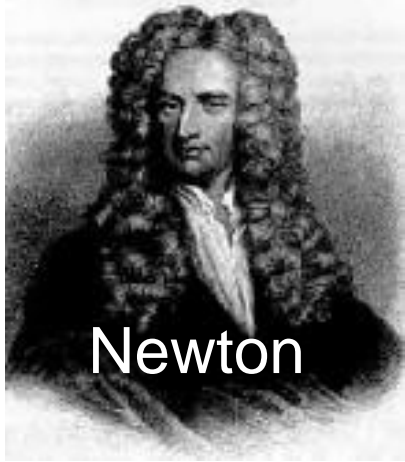
Find the direction of  $\vec{A}$  at the perigee.



Direction of the LRL vector is: *focus to perigee.*

$\vec{A}$  : Must remain constant  
 – *no matter where the planet is!*

This is precisely what FIXES the orbit!



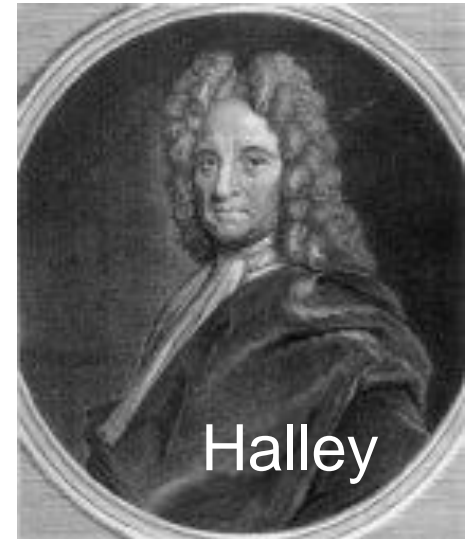
Newton

Force (per unit mass):  $\frac{d\vec{v}}{dt} = -\frac{\kappa}{\rho^2} \hat{e}_\rho$

Newton told us (but first, to Halley)!

Given the constancy related to the conservation of the LRL vector, could you have discovered the law of gravity?

If you did that, wouldn't you have discovered a law of nature?



Halley

## Symmetry & Conservation Principles!

“It is now natural for us to try to derive the laws of nature and to test their validity by means of the laws of invariance, rather than to derive the laws of invariance from what we believe to be the laws of nature.” - Eugene Wigner



PCD\_STiCM

**Eugene  
Paul  
Wigner  
(1902-1995)**

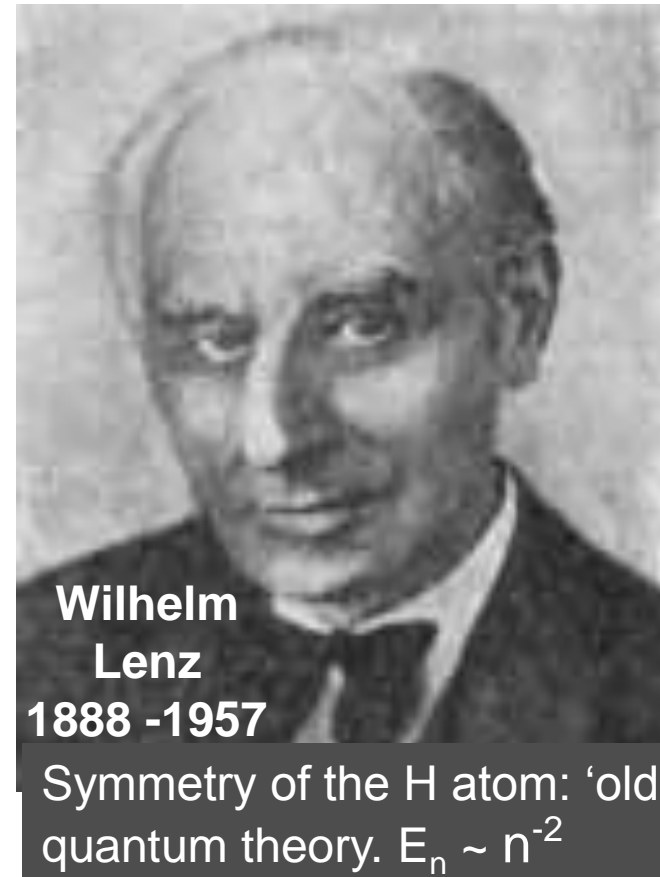
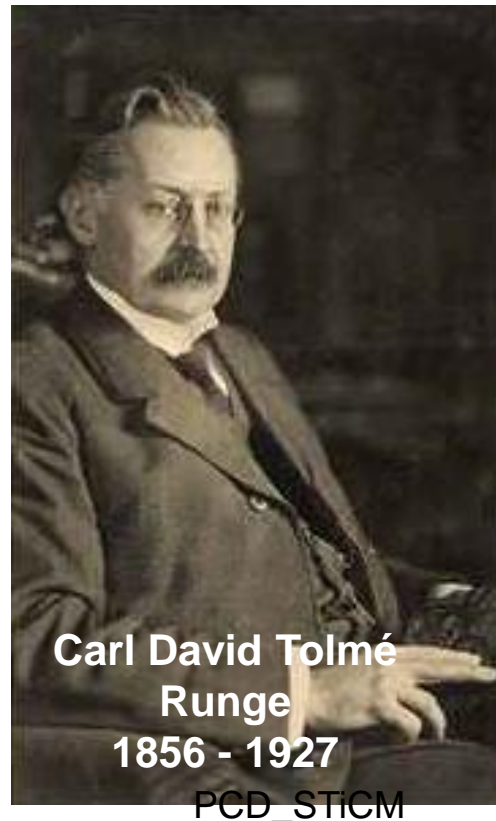
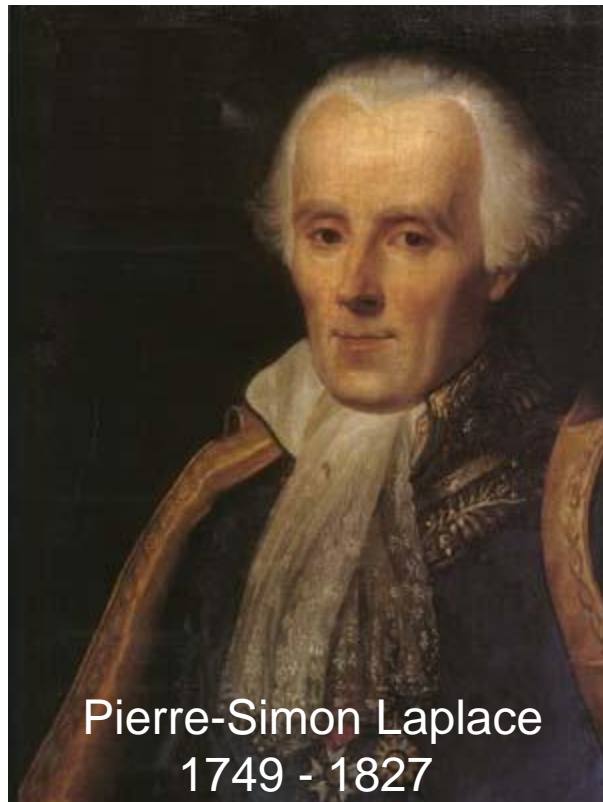


$$\vec{A} = \vec{v} \times \vec{H} - \kappa \hat{e}_\rho$$

Laplace Runge Lenz Vector :

constant for a strict  $\frac{1}{r}$  potential.

Conservation law associated with '*dynamical / accidental*' symmetry.



# Further reading:

- Symmetry & Conservation laws play an important role in understanding the very frontiers of Physics.
- The implications go as far as testing the ‘standard’ model of physics, and exploring if there is any physics beyond the standards model.

Do visit:

Feynman's Messenger Lectures Online

AKA Project Tuva

[http://www.fotuva.org/news/project\\_tuva.html](http://www.fotuva.org/news/project_tuva.html)

## Useful references on 'Symmetry & Conservation Laws'

P. C. Deshmukh and Shyamala Venkataraman

*Obtaining Conservation Principles from Laws of Nature -- and the other way around!*

Bulletin of the Indian Association of Physics Teachers, Vol. 3, 143-148 (2011)

P. C. Deshmukh and J. Libby

*(a) Symmetry Principles and Conservation Laws in Atomic and Subatomic Physics -1*

*Resonance*, 15, 832 (2010)

*b) Symmetry Principles and Conservation Laws in Atomic and Subatomic Physics -2*

*Resonance*, 15, 926 (2010)

Continuous Symmetries - Translation, Rotation

Dynamical Symmetries - LRL, Fock Symmetry  
SO(4)

Discrete Symmetries

- P : Parity
- C : Charge Conjugation
- T : Time Reversal

Lorentz symmetry: associated with the PCT symmetry.

PCT theorem (Wolfgang Pauli)

No experiment has revealed any violation of PCT symmetry.

This is predicted by the Standard Model of particle physics.

## The Standard Model today

### Elementary particles

	First family	Second family	Third family
Leptons	electron neutrino electron	muon neutrino muon	tau neutrino tau
Quarks	up down	charm strange	top bottom

### Forces

### Messenger particles

electromagnetic force	photon
weak force	W, Z
strong force	gluons

Higgs?

*The 'standard model' unifies all the fundamental building blocks of matter, and three of the four fundamental forces.*

*To complete the Model a new particle is needed – the Higgs Boson – that the physics community hopes to find in the new built accelerator LHC at CERN in Geneva.*

## 2008 Nobel Prize in Physics

"for the discovery of the mechanism of **spontaneous broken symmetry** in subatomic physics"



Yoichiro Nambu  
 $\frac{1}{2}$  prize  
 Enrico Fermi Institute,  
 Chicago, USA

"for the discovery of the origin of the **broken symmetry which predicts the existence of at least three families of quarks in nature**"



Makoto Kobayashi  
 $\frac{1}{4}$  prize  
 High Energy  
 Accelerator



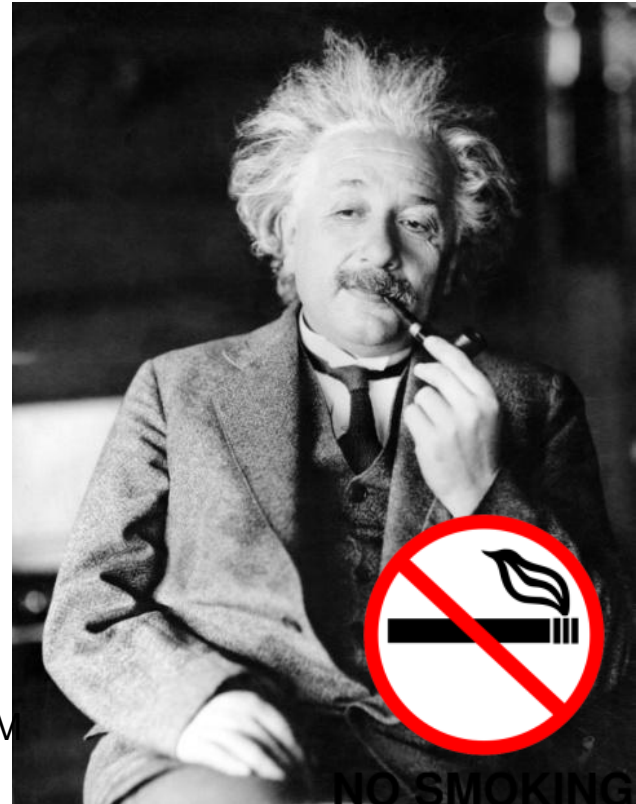
Toshihide Maskawa  
 $\frac{1}{4}$  prize  
 Kyoto Sangyo Univ,  
 Kyoto Japan

PCD\_Research Organization,  
 Tsukuba, Japan



'every symmetry in nature yields  
a conservation law  
and conversely,  
every conservation law reveals  
an underlying symmetry'.

## Noether's theorem





“In the judgment of the most competent living mathematicians, Fräulein Noether was the most significant creative mathematical genius thus far produced since the higher education of women began.”

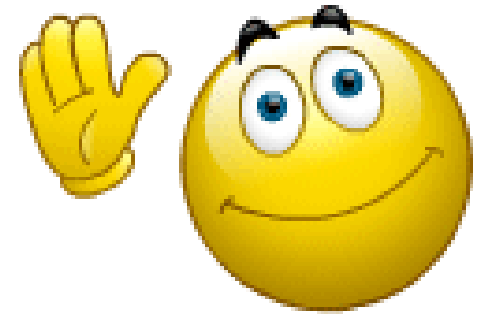
“In the realm of algebra, ... , she discovered methods which have proved of enormous importance .... ”

“..... Her unselfish, significant work over a period of many years was rewarded by the new rulers of Germany with a dismissal, which cost her the means of maintaining her simple life and the opportunity to carry on her mathematical studies.....”

ALBERT EINSTEIN. Princeton University, May 1, 1935.

New York Times May 5, 1935. *Excerpts*

We will take a Break...  
.....*Any questions ?*



[pcd@physics.iitm.ac.in](mailto:pcd@physics.iitm.ac.in)

Next: Unit 5: Inertial and non-inertial reference frames.

Moving coordinate systems. Pseudo forces.

Inertial and non-inertial reference frames.

‘Deterministic’ cause-effect relations in inertial frame,  
and their *modifications* in a *non-inertial* frame.